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No-tension bodies: a reinforcement problem

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Abstract

In this work we show that the framework put forward by Lucchesi, Silhavy and Zani [8] to study the equilibrium configurations of panels made of no-tension material can be easily extended to the case of a no-tension material with a reinforcing tensile resistant unidimensional material. This kind of bodies could be used to describe reinforced concrete structures. By solving the equilibrium equations we find a family of solutions each of which is characterized by a singular curve where the stress in the no-tension material concentrates. We show that among these, the curve that minimizes the maximum stress resembles the line tension found experimentally on reinforced concrete beams.

Keywords:

No-tension materials, singular stresses, equilibrated tensor fields, reinforced concrete beams

1. Introduction

In this work we show that the framework put forward by Lucchesi, Silhavy and Zani [8] to study the equilibrium configurations of panels made of no-tension material, [5, 6, 7], can be easily extended to the case of a no-tension material with a reinforcing tensile resistant unidimensional material. This kind of bodies could be used to describe reinforced concrete structures.

In [8, 10] Lucchesi, Silhavy and Zani look for stress fields that equilibrate the applied loads and are negative semi-definite, so to accommodate the incapability of the material to withstand traction. To simplify the problem they use tensor valued measures to describe the stress field; more precisely, they consider stresses that are tensor valued measures with a divergence which is also a measure, see also [4]. Within this framework the stress field may be singular on some curve, C_u , to be determined from the equilibrium equations; highly localized stress distributions have been experimentally observed in [2, 3]. The theory for these generalized stresses has been developed by Lucchesi *et al.* in [8, 9, 10, 11, 12]; see also [1]. In particular, the balance of forces is postulated only in a weak

form that allows to take into account the singularities of the stress in a simple and direct way.

To model a no-tension body with a unidimensional reinforcement we prescribe, in the reference configuration, a fixed curve C_r representing the region occupied by the reinforcement. On this curve the stress field will be allowed to be singular and to be positive definite in order to model the fact that the reinforcement could support traction forces.

While in the reinforcement the only unknown is the stress, since C_r is a priori given, in the singular curve within the no-tension material the unknowns are the stress and the curve C_u itself. Besides these, also the density with respect to the Lebesgue measure of the stress field in the no-tension material is unknown.

More precisely, we deal with a body occupying a region U that is divided by a singular “curve” $C = C_u \cup C_r$ and the stress \mathbb{T} is a measure that is the sum of an absolutely continuous part w.r.t. the Lebesgue measure on $U \setminus C$, with density \mathbb{T}_a , and a measure concentrated on C . Denoting by \mathbb{T}_r and \mathbb{T}_u the density of the measures concentrated on C_r and C_u , respectively, we require that \mathbb{T}_a and \mathbb{T}_u are negative semi-definite, since they represent the stress in the no-tension material while \mathbb{T}_r could be positive. Therefore, the unknowns of the problem are \mathbb{T}_a , \mathbb{T}_r , \mathbb{T}_u and C_u .

In Section 2, after recalling one of the main results of [8], we derive the equilibrium equations from the balance equations in the weak form for the case of a rectangular panel with a straight horizontal reinforcement. The choice of this configuration is motivated by the fact that it describes the geometry of reinforced concrete beams. In Section 3 we study the equilibrium problem for the structure subjected to a uniform load on the top of the panel and clamped on two intervals at its basis. Since we are considering only the equilibrium equations, we find that the solution is not unique. In fact, we find a family of solutions each of which is characterized by a singular curve C_u . We show that the curve that minimizes the maximum stress in the no-tension material, which we call *optimal singular curve*, resembles the line tension found experimentally on reinforced concrete beams.

2. Equilibrated tensor fields and balance equations

In this section we briefly recall the basic notion of equilibrated tensor field and the corresponding balance equations using the notation of [8], see also [10, 14]. To the same paper, and to the references therein, we refer for a complete and detailed presentation.

Let U be an open subset of \mathbb{R}^n and ∂U its topological boundary. Let V be a finite-dimensional real inner product space. We denote by $M(U, V)$ and $M(\partial U, V)$ the set of V -valued Borel measures supported on U and ∂U , respectively. By Lin we denote the space of all linear transformations (tensors) from \mathbb{R}^n into \mathbb{R}^n with the Euclidean inner product.

Definition 1. *A tensor-valued measure $T \in M(U, \text{Lin})$ is said to be an equilibrated tensor field if there exist measures (actually unique) $b_0 \in M(U, \mathbb{R}^n)$ and*

$t_0 \in M(\partial U, \mathbb{R}^n)$ such that

$$\int_U \nabla \varphi \cdot dT = \int_U \varphi \cdot db_0 + \int_{\partial U} \varphi \cdot dt_0 \quad (1)$$

for each $\varphi \in C_c^\infty(\mathbb{R}^n; \mathbb{R}^n)$. The pair (b_0, t_0) is called the load corresponding to T .

It follows that, the distributional divergence of an equilibrated tensor field T is a vector measure $\operatorname{div} T \in M(U; \mathbb{R}^n)$. By using (1) we can see that the map

$$\langle N(T), \varphi \rangle := \int_U \nabla \varphi \cdot dT + \int_U \varphi \cdot d\operatorname{div} T, \quad \varphi \in C_c^\infty(\mathbb{R}^n, \mathbb{R}^n) \quad (2)$$

is a measure concentrated on ∂U , which is called the *normal trace* of T at the boundary; hence

$$\langle N(T), \varphi \rangle = \int_{\partial U} \varphi \cdot dN(T)$$

and the following Green's formula

$$\int_U \nabla \varphi \cdot dT = - \int_U \varphi \cdot d\operatorname{div} T + \int_{\partial U} \varphi \cdot dN(T) \quad (3)$$

holds for every $\varphi \in C_c^\infty(\mathbb{R}^n, \mathbb{R}^n)$.

By comparing (1) and (3) we obtain that any equilibrated tensor field satisfies the balance equations

$$\begin{cases} -\operatorname{div} T = b_0, \\ N(T) = t_0. \end{cases} \quad (4)$$

Actually, this is the set of equilibrium equations for a continuous body under the action of a body force given by a prescribed measure $b_0 \in M(U, \mathbb{R}^n)$ and a boundary traction given by a prescribed measure $t_0 \in M(\partial U, \mathbb{R}^n)$. In particular, if b_0 is absolutely continuous with respect to the Lebesgue measure then $\operatorname{div} T$ must be absolutely continuous as well.

2.1. A reinforced panel

In [8], Lucchesi, Šilavý and Zani, after having developed the theory of equilibrated stress fields for no-tension bodies, have studied several two dimensional equilibrium problems for multi-rectangular panels.

Inspired by their work, we study the statics of a two dimensional rectangular panel made of no-tension material and reinforced by means of a straight uni-dimensional continuum capable to resist also to traction forces. This system models, for instance, a reinforced concrete beam.

We denote by U the rectangular region occupied by the panel in its reference configuration and by $C_r \subset U$ the uni-dimensional straight line occupied by the reinforcing material. We assume C_r to be parallel to the basis of U . Since the panel is made of no-tension material, the stress in $U \setminus C_r$ is assumed to be symmetric and negative semidefinite.

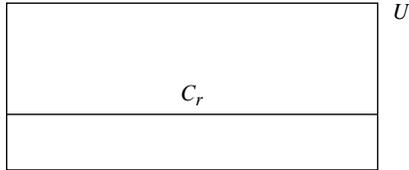


Figure 1: a concrete panel reinforced by a straight unidimensional continuum made of tensile resistant material

In the next section we consider an equilibrium problem under prescribed loads and constraints and look for solutions among equilibrated tensor fields \mathbb{T} , which are actually solutions to the equilibrium equations (4).

Following the ideas of [8], in this paper we are going to search only special solutions of (4) by restricting the set of admissible stress tensor fields to those \mathbb{T} whose singularities concentrate on C_r and along a simple piecewise smooth curve C_u (like for instance in Figure 2 where $C_u = C_1 \cup C_2 \cup C_3$) with endpoints h_1 and h_2 on the boundary of U and disjoint from the endpoints e_1 and e_2 of C_r .

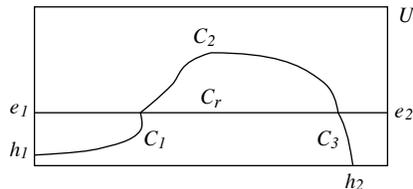


Figure 2: we allow \mathbb{T} to be singular on C_r and along an unknown simple piecewise smooth curve $C_u = C_1 \cup C_2 \cup C_3$

This means that the stress field \mathbb{T} is assumed to be the sum of a measure absolutely continuous with respect to the Lebesgue's measure with a smooth density \mathbb{T}_a in $U \setminus (C_u \cup C_r)$ which has a continuous extension, again denoted by \mathbb{T}_a , with \mathcal{L}^2 -integrable derivative, to the closure of any connected component of $U \setminus (C_u \cup C_r)$ (hence, in particular, \mathcal{H}^1 -integrable on ∂U) and two measures concentrated on C_r and C_u whose densities are piecewise smooth superficial tensor fields \mathbb{T}_r and \mathbb{T}_u , respectively, that is

$$\mathbb{T}_r = \sigma_r(s)t_r(s) \otimes t_r(s), \quad \mathbb{T}_u = \sigma_u(s)t_u(s) \otimes t_u(s), \quad (5)$$

where σ_r and σ_u are piecewise smooth scalar fields, respectively, on C_r and C_u (that is they are allowed to jump only on the intersection points of the two curves with finite right and left limits in such points and have \mathcal{H}^1 -integrable derivative), s is the arclength and $t_r(s)$, $t_u(s)$ denote the respective tangent

unit vectors. Summarizing, the unknowns of the problem are the curve C_u , the tensor field \mathbb{T}_a and the scalar fields σ_r and σ_u .

The admissible stress tensor fields are then of the form

$$\mathbb{T} := \mathbb{T}_a \mathcal{L}^2 \llbracket U + \mathbb{T}_r \mathcal{H}^1 \llbracket C_r + \mathbb{T}_u \mathcal{H}^1 \llbracket C_u \quad (6)$$

and, as shown in [8, Proposition 1], they are equilibrated. Therefore, once having defined the applied loads, the equilibrium equations are given by (4). To make them more explicit we compute $\operatorname{div} \mathbb{T}$ and $\mathbf{N}(\mathbb{T})$. It will be shown that

$$\begin{aligned} \operatorname{div} \mathbb{T} &= \operatorname{div} \mathbb{T}_a \mathcal{L}^2 \llbracket U \\ &+ \left(\frac{d}{ds} (\sigma_r \mathbf{t}_r) - [\mathbb{T}_a] \mathbf{n}_r \right) \mathcal{H}^1 \llbracket C_r \\ &+ \left(\frac{d}{ds} (\sigma_u \mathbf{t}_u) - [\mathbb{T}_a] \mathbf{n}_u \right) \mathcal{H}^1 \llbracket C_u \\ &+ \sum_{j=1}^p [\sigma_r] \mathbf{t}_r \delta_{c_j} + \sum_{j=1}^p [\sigma_u \mathbf{t}_u] \delta_{c_j} \end{aligned} \quad (7)$$

and

$$\mathbf{N}(\mathbb{T}) = \mathbb{T}_a \mathbf{m} \mathcal{H}^1 \llbracket \partial U + \sigma_r \mathbf{t}_r (\delta_{e_2} - \delta_{e_1}) + \sigma_u \mathbf{t}_u (\delta_{h_2} - \delta_{h_1}) \quad (8)$$

where $(\mathbf{t}_r, \mathbf{n}_r)$ and $(\mathbf{t}_u, \mathbf{n}_u)$ are unit tangent and normal vectors to C_r and C_u , \mathbf{m} is the outer normal to ∂U and c_j ($j = 1, \dots, p$) are the intersection points between the two curves. Moreover, $[\mathbb{T}_a]$ denotes the jump of \mathbb{T}_a across the curves, while $[\sigma_r \mathbf{t}_r](P)$ and $[\sigma_u \mathbf{t}_u](P)$ are the jumps of σ_r and σ_u at the point P along C_r and C_u . The jumps are evaluated according to the orientation defined, either by the normal or the tangent vectors. Hence, for instance, $[\sigma_u \mathbf{t}_u] \delta_{c_1}$ simply means $[\sigma_u \mathbf{t}_u](c_1) := \sigma_u(s(c_1)^+) \mathbf{t}_u(s(c_1)^+) - \sigma_u(s(c_1)^-) \mathbf{t}_u(s(c_1)^-)$, where $+$ and $-$ denote the right and left limits referred to the chosen parametrization s .

To simplify the computations, we confine ourselves to the case $p = 2$; that is $C_r \cap C_u = \{c_1, c_2\} \subset U$. In this case there are two possible situations: one in which the two endpoints of C_u are both below C_r as in Figure 2, and the other in which they are both above. To fix ideas we suppose to be in the first situation.

The domain U turns out to be divided in 5 parts called U_i , $i = 1, 2, 3, 4, 5$, according to Figure 3.

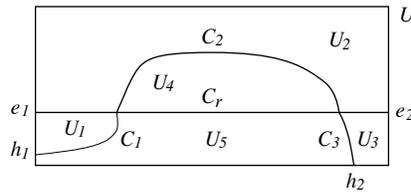


Figure 3: the five regions U_i , $i = 1, 2, 3, 4, 5$

Recalling the expression (6) of \mathbb{T} , for any $\varphi \in C_c(\mathbb{R}^2; \mathbb{R}^2)$ we have

$$\begin{aligned} \int_U \nabla \varphi \cdot d\mathbb{T} &= \int_U \nabla \varphi \cdot \mathbb{T}_a d\mathcal{L}^2 + \int_{C_r} \nabla \varphi \cdot \mathbb{T}_r d\mathcal{H}^1 \\ &\quad + \int_{C_u} \nabla \varphi \cdot \mathbb{T}_u d\mathcal{H}^1. \end{aligned} \quad (9)$$

After denoting by \mathbb{T}_a^i the trace of \mathbb{T}_a on ∂U_i and \mathbf{n}^i the corresponding outer normal, the first term on the right-hand side of (9) becomes

$$\begin{aligned} \int_U \nabla \varphi \cdot \mathbb{T}_a d\mathcal{L}^2 &= \\ &= - \sum_{i=1}^5 \int_{U_i} \varphi \cdot \operatorname{div} \mathbb{T}_a dx + \sum_{i=1}^5 \int_{\partial U_i} \varphi \cdot \mathbb{T}_a^i \mathbf{n}^i ds \\ &= - \int_U \varphi \cdot \operatorname{div} \mathbb{T}_a dx + \int_{\partial U} \varphi \cdot \mathbb{T}_a \mathbf{m} ds \\ &\quad + \sum_{i,j=1, i < j}^5 \int_{\partial U_i \cap \partial U_j} \varphi \cdot (\mathbb{T}_a^i - \mathbb{T}_a^j) \mathbf{n}^i ds \\ &= - \int_U \varphi \cdot \operatorname{div} \mathbb{T}_a dx + \int_{\partial U} \varphi \cdot \mathbb{T}_a \mathbf{m} ds \\ &\quad + \int_{C_r} \varphi \cdot [\mathbb{T}_a] \mathbf{n}_r ds + \int_{C_u} \varphi \cdot [\mathbb{T}_a] \mathbf{n}_u ds, \end{aligned}$$

where

$$[\mathbb{T}_a] \mathbf{n}_r = \begin{cases} (\mathbb{T}_a^1 - \mathbb{T}_a^2) \mathbf{n}^1 & \text{on } \partial U_1 \cap \partial U_2, \\ (\mathbb{T}_a^2 - \mathbb{T}_a^3) \mathbf{n}^2 & \text{on } \partial U_2 \cap \partial U_3, \\ (\mathbb{T}_a^4 - \mathbb{T}_a^5) \mathbf{n}^4 & \text{on } \partial U_4 \cap \partial U_5, \end{cases}$$

and

$$[\mathbb{T}_a] \mathbf{n}_u = \begin{cases} (\mathbb{T}_a^1 - \mathbb{T}_a^5) \mathbf{n}^1 & \text{on } \partial U_1 \cap \partial U_5 = C_1, \\ (\mathbb{T}_a^2 - \mathbb{T}_a^4) \mathbf{n}^2 & \text{on } \partial U_2 \cap \partial U_4 = C_2, \\ (\mathbb{T}_a^3 - \mathbb{T}_a^5) \mathbf{n}^3 & \text{on } \partial U_3 \cap \partial U_5 = C_3. \end{cases} \quad (10)$$

To make the computation of the second term on the right-hand side of (9) easier, we introduce a positive orientation on the curve C_r in which e_1 is the initial endpoint (hence e_2 is the final endpoint). By using the fact that $(\mathbf{t}_r \otimes \mathbf{t}_r) \cdot \nabla(\varphi \circ \gamma_r) = \mathbf{t}_r \cdot \frac{d}{ds}(\varphi \circ \gamma_r)$ where $\gamma_r(s)$ is any regular parametrization of C_r and

recalling the expression (5) of \mathbb{T}_r , we have

$$\begin{aligned}
& \int_{C_r} \nabla \varphi \cdot \mathbb{T}_r d\mathcal{H}^1 = \\
& = \int_{\partial U_1 \cap \partial U_2} \sigma_r(\mathbf{t}_r \otimes \mathbf{t}_r) \cdot \nabla \varphi ds + \int_{\partial U_2 \cap \partial U_3} \sigma_r(\mathbf{t}_r \otimes \mathbf{t}_r) \cdot \nabla \varphi ds \\
& \quad + \int_{\partial U_4 \cap \partial U_5} \sigma_r(\mathbf{t}_r \otimes \mathbf{t}_r) \cdot \nabla \varphi ds \\
& = - \int_{C_r} \frac{d}{ds} (\sigma_r \mathbf{t}_r) \cdot \varphi ds \\
& \quad + \sigma_r(c_1^-) \mathbf{t}_r(c_1) \cdot \varphi(c_1) - \sigma_r(e_1) \mathbf{t}_r(e_1) \cdot \varphi(e_1) \\
& \quad + \sigma_r(c_2^-) \mathbf{t}_r(c_2) \cdot \varphi(c_2) - \sigma_r(c_1^+) \mathbf{t}_r(c_1) \cdot \varphi(c_1) \\
& \quad + \sigma_r(e_2) \mathbf{t}_r(e_2) \cdot \varphi(e_2) - \sigma_r(c_2^+) \mathbf{t}_r(c_2) \cdot \varphi(c_2) \\
& = - \int_{C_r} \frac{d}{ds} (\sigma_r \mathbf{t}_r) \cdot \varphi ds - \int [\sigma_r] \mathbf{t}_r \cdot \varphi d(\delta_{c_1} + \delta_{c_2}) \\
& \quad + \int \sigma_r \mathbf{t}_r \cdot \varphi d(\delta_{e_2} - \delta_{e_1}).
\end{aligned} \tag{11}$$

Analogously, having fixed on C_u a positive orientation which goes from h_1 to h_2 , we have

$$\begin{aligned}
& \int_{C_u} \nabla \varphi \cdot \mathbb{T}_u d\mathcal{H}^1 = \\
& = - \int_{C_u} \frac{d}{ds} (\sigma_u \mathbf{t}_u) \cdot \varphi ds - \int [\sigma_u \mathbf{t}_u] \cdot \varphi d(\delta_{c_1} + \delta_{c_2}) \\
& \quad + \int \sigma_u \mathbf{t}_u \cdot \varphi d(\delta_{h_2} - \delta_{h_1}).
\end{aligned} \tag{12}$$

By putting all together we find

$$\begin{aligned}
\int_U \nabla \varphi \cdot d\mathbb{T} & = - \int_U \varphi \cdot \operatorname{div} \mathbb{T}_a dx + \int_{\partial U} \varphi \cdot \mathbb{T}_a m d\mathcal{H}^1 \\
& \quad + \int_{C_r} ([\mathbb{T}_a] \mathbf{n}_r - \frac{d}{ds} (\sigma_r \mathbf{t}_r)) \cdot \varphi d\mathcal{H}^1 \\
& \quad + \int_{C_u} ([\mathbb{T}_a] \mathbf{n}_u - \frac{d}{ds} (\sigma_u \mathbf{t}_u)) \cdot \varphi d\mathcal{H}^1 \\
& \quad - \int ([\sigma_r] \mathbf{t}_r + [\sigma_u \mathbf{t}_u]) \cdot \varphi d(\delta_{c_1} + \delta_{c_2}) \\
& \quad + \sigma_r \mathbf{t}_r \cdot \varphi d(\delta_{e_2} - \delta_{e_1}) \\
& \quad + \int \sigma_u \mathbf{t}_u \cdot \varphi d(\delta_{h_2} - \delta_{h_1}).
\end{aligned}$$

The claimed expression for $\operatorname{div} \mathbb{T}$ is then obtained by taking test functions $\varphi \in$

$C_c^\infty(U; \mathbb{R}^2)$ and using the fact that, within this choice,

$$\int_U \varphi \cdot d\operatorname{div}\mathbb{T} = - \int_U \nabla\varphi \cdot d\mathbb{T}.$$

After that, $\mathbb{N}(\mathbb{T})$ is easily computed by means of (2).

3. The equilibrium problem

In this section we assume the panel to be subjected to boundary loads only and to be clamped on two regions V_1 and V_2 of width d contained in the lower basis.

It is useful to introduce an orthogonal coordinate system (x, y) with the origin in the middle point of the lower side of the panel, with the x axis pointing right and the y axis pointing upward (see Figure 4); let (\bar{e}_1, \bar{e}_2) be the associated canonical basis. In this reference we take $U = (-b/2, b/2) \times (0, h)$, $V_1 = (-b/2, -b/2 + d) \times \{0\}$, $V_2 = (b/2 - d, b/2) \times \{0\}$ with $b, h > 0$ and $d \in (0, b/2)$.

We assume that the system is subjected to a vertical load, $-p_0\bar{e}_2$, distributed on its upper side, $y = h$. The couple of measures describing the loads is then given by

$$b_0 = 0, \quad t_0 = -p_0\bar{e}_2[\Gamma + \Phi[V_1 \cup V_2]$$

where $\Gamma = (-b/2, b/2) \times \{h\}$ is the upper side of the panel and V_1 and V_2 are the subsets of the lateral boundary of U in which the body is clamped and $\Phi[V_1 \cup V_2]$ is a vector valued measure representing the reaction of the constraint and which is a-priori unknown.

Thanks to (7) and (8), the equilibrium equations (4) rewrite

$$\left\{ \begin{array}{l} \operatorname{div}\mathbb{T}_a \mathcal{L}^2[U + (\frac{d}{ds}(\sigma_r t_r) - [\mathbb{T}_a]n_r) \mathcal{H}^1[C_r + \\ + (\frac{d}{ds}(\sigma_u t_u) - [\mathbb{T}_a]n_u) \mathcal{H}^1[C_u \\ + \sum_{j=1}^2 [\sigma_r] t_r \delta_{c_j} + \sum_{j=1}^2 [\sigma_u t_u] \delta_{c_j} = 0, \\ \mathbb{T}_a m \mathcal{H}^1[\partial U + \sigma_r t_r (\delta_{e_2} - \delta_{e_1}) + \sigma_u t_u (\delta_{h_2} - \delta_{h_1}) = \\ = -p_0\bar{e}_2[\Gamma + \Phi[V_1 \cup V_2], \end{array} \right.$$

that is

$$\left\{ \begin{array}{ll} \operatorname{div} \mathbb{T}_a = 0 & \mathcal{L}^2 - \text{a.e. in } U, \\ \frac{d}{ds}(\sigma_r t_r) - [\mathbb{T}_a] n_r = 0 & \mathcal{H}^1 - \text{a.e. on } C_r, \\ \frac{d}{ds}(\sigma_u t_u) - [\mathbb{T}_a] n_u = 0 & \mathcal{H}^1 - \text{a.e. on } C_u, \\ [\sigma_r] t_r + [\sigma_u] t_u = 0 & \text{in } c_1, c_2, \\ \mathbb{T}_a m = -p_0 \bar{e}_2 & \mathcal{H}^1 - \text{a.e. on } \Gamma, \\ \mathbb{T}_a m = 0 & \mathcal{H}^1 - \text{a.e. on } \partial U \setminus (\Gamma \cup V_1 \cup V_2), \\ \sigma_r t_r(e_2) = \sigma_r t_r(e_1) = 0, \\ \mathbb{T}_a m \mathcal{H}^1[V_1 \cup V_2 + \sigma_u t_u(\delta_{h_2} - \delta_{h_1}) = \Phi[V_1 \cup V_2. \end{array} \right. \quad (13)$$

Let us remark that the last equation is satisfied if the constraint is able to produce a reaction $\Phi[V_1 \cup V_2]$ as prescribed by the left-hand side and we assume, from now on, that this is true whenever $h_1, h_2 \in V_1 \cup V_2$. If, on the contrary, $h_i \notin V_1 \cup V_2$, $i = 1, 2$, then the equation is satisfied only if $\sigma_u(h_i) = 0$.

Since the geometry of the domain and the applied loads are symmetric with respect to the axis $x = 0$, we look for solutions with the same kind of symmetry. In particular, this implies that if $h_1 \in V_1$ then $h_2 \in V_2$. Moreover, the unknown curve C_u must be union of three smooth curves C_1 , C_2 and C_3 connecting the points h_1 , c_1 , c_2 and h_2 as in Figure 4, C_2 must be symmetric with respect to the axis $x = 0$ and C_3 must be obtained by reflecting C_1 through to the axis y .

Denoting by $a \in (0, h)$ the distance between the reinforcing line C_r and the bottom of U , and by $\mu \in [0, b/2)$ the distance of the points c_1 and c_2 from the axis y , we have that $c_1 = (-\mu, a)$ and $c_2 = (\mu, a)$.

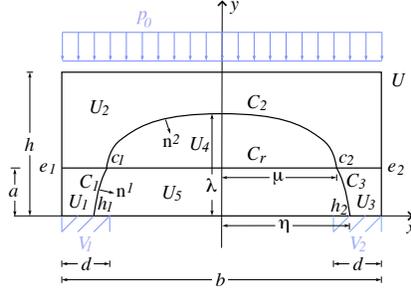


Figure 4: the loaded system

For computational convenience we look for solutions C_i that are graphics

with parametric representations

$$\begin{aligned} C_1 &= \{(w_1(y), y) : y \in [0, a]\}, \quad w_1 \in C^1([0, a]), \\ C_2 &= \{(x, w_2(x)) : x \in [-\mu, \mu]\}, \quad w_2 \in C^1([-\mu, \mu]), \\ C_3 &= \{(-w_1(y), y) : y \in [0, a]\}. \end{aligned} \quad (14)$$

Of course $C_r = \{(x, a) : x \in [-b/2, b/2]\}$.

With $J_i := \sqrt{1 + |w'_i|^2}$, the tangent unit vectors to C_1 , C_2 and C_r are, respectively, $t^1 = J_1^{-1}(w'_1 \bar{e}_1 + \bar{e}_2)$, $t^2 = J_2^{-1}(\bar{e}_1 + w'_2 \bar{e}_2)$, and $t_r = (1, 0)$. Then we have $n^1 = J_1^{-1}(\bar{e}_1 - w'_1 \bar{e}_2)$ and $n^2 = J_2^{-1}(w'_2 \bar{e}_1 - \bar{e}_2)$.

Following the ideas of [8], we observe that the first, the fifth and the sixth equations in (13) are satisfied if \mathbb{T}_a is given by

$$\mathbb{T}_a := \begin{cases} -p_0 \bar{e}_2 \otimes \bar{e}_2 & \text{in } U_1 \cup U_2 \cup U_3, \\ 0 & \text{in } U_4 \cup U_5. \end{cases} \quad (15)$$

We now study the remaining equations in (13).

Second and seventh equation. Since $[T_a] = 0$ \mathcal{H}^1 -a.e. across C_r and since t_r is a constant vector, then the second equation is equivalent to

$$\sigma_r = \text{locally constant on } C_r \setminus \{c_1, c_2\}.$$

Using the boundary conditions prescribed by the seventh equation $\sigma_r t_r(e_2) = \sigma_r t_r(e_1) = 0$ then we conclude that

Proposition 1. σ_r is zero on the segments $\overline{e_1 c_1}$ and $\overline{c_2 e_2}$ and is equal to a constant $\bar{\sigma}_r$ on the segment $\overline{c_1 c_2}$.

Third equation. On the curves C_1 and C_2 , by (10), the third equation writes

$$\begin{cases} \frac{d}{ds}(\sigma_1 t^1) - (\mathbb{T}_a^1 - \mathbb{T}_a^5) n^1 = 0 & \text{on } C_1, \\ \frac{d}{ds}(\sigma_2 t^2) - (\mathbb{T}_a^2 - \mathbb{T}_a^4) n^2 = 0 & \text{on } C_2, \end{cases}$$

where $\sigma_\alpha = \sigma_u|_{C_\alpha}$. From (15) and the explicit form of the normals, we find

$$(\mathbb{T}_a^1 - \mathbb{T}_a^5) n^1 = J_1^{-1} p_0 w'_1 \bar{e}_2, \quad (\mathbb{T}_a^2 - \mathbb{T}_a^4) n^2 = J_2^{-1} p_0 \bar{e}_2,$$

and

$$\frac{d}{ds}(\sigma_\alpha t^\alpha) = J_\alpha^{-1} \frac{d}{dx}(\sigma_\alpha t^\alpha).$$

Setting

$$\beta_\alpha := \frac{\sigma_\alpha}{J_\alpha}, \quad \alpha = 1, 2, \quad (16)$$

the third equation rewrites as

$$\boxed{\begin{cases} (\beta_1 w_1')' = 0, \\ \beta_1' - p_0 w_1' = 0, \\ \beta_2' = 0, \\ (\beta_2 w_2')' - p_0 = 0, \\ \sigma_1 = \beta_1 \sqrt{1 + |w_1'|^2}, \\ \sigma_2 = \beta_2 \sqrt{1 + |w_2'|^2}. \end{cases}} \quad (17)$$

Fourth equation. The fourth equation in the point c_1 is

$$[\sigma_r]t_r(c_1) + [\sigma_u]t_u(c_1) = 0.$$

By Proposition 1 the above equation rewrites

$$\sigma_r(c_1^+) \bar{e}_1 + \sigma_u(c_1^+) t_u(c_1^+) - \sigma_u(c_1^-) t_u(c_1^-) = 0,$$

that is

$$\bar{\sigma}_r \bar{e}_1 + \sigma_2(0) t^2(0) - \sigma_1(a) t^1(a) = 0.$$

With the explicit expressions for $t^2(0)$ and $t^1(a)$ we obtain that the fourth equation becomes

$$\boxed{\begin{cases} \bar{\sigma}_r + \beta_2(-\mu) - \sigma_1(a) w_1'(a) = 0, \\ \beta_2(-\mu) w_2'(-\mu) - \beta_1(a) = 0. \end{cases}} \quad (18)$$

To the set of differential equations (17) and boundary conditions (18) we can add the additional boundary condition

$$w_2(-\mu) = a. \quad (19)$$

It is also useful to set

$$w_2(0) =: \lambda > a \quad (20)$$

and remark that

$$w_1(0) = -\eta, \quad w_1(a) = -\mu, \quad w_2'(0) = 0 \quad (21)$$

since w_2 is a smooth even function. Above η denotes the distance of the points h_1 and h_2 from the origin.

We now solve for C_2 and σ_2 .

From (17)₃ we obtain that β_2 is constant; moreover $\beta_2 \neq 0$ since otherwise (17)₄ would imply $p_0 = 0$. Integrating equation (17)₄ and using (21)₃ we get

$$\beta_2 w_2'(x) = p_0 x.$$

Integrating again and using (20) we obtain

$$w_2(x) = \lambda + \frac{p_0}{2\beta_2} x^2,$$

and the boundary condition (19) gives

$$\beta_2 = -\frac{\mu^2 p_0}{2(\lambda - a)}. \quad (22)$$

Therefore

$$w_2(x) = \lambda - \frac{\lambda - a}{\mu^2} x^2, \quad x \in [-\mu, \mu]$$

and (17)₆ gives

$$\sigma_2(x) = -p_0 \sqrt{\frac{\mu^4}{4(\lambda - a)^2} + x^2}, \quad x \in [-\mu, \mu]. \quad (23)$$

We now solve for C_1 and σ_1 .

Integrating (17)₂ we get

$$\beta_1 - p_0 w_1 = c,$$

with c constant. Using (18)₂ and (21)₂ and the expressions of w_2 and σ_2 computed before we find

$$c = \beta_1(a) - p_0 w_1(a) = \beta_2 w_2'(-\mu) + p_0 \mu = 0,$$

hence

$$\beta_1 = p_0 w_1. \quad (24)$$

With (24), from equation (17)₁ we obtain

$$\frac{p_0}{2} w_1(y)^2 = ky + d,$$

where k and d are constants determined by the boundary conditions. In fact, by using (21)₁ and (21)₂ we find

$$k = \frac{p_0}{2a}(\mu^2 - \eta^2), \quad d = \frac{p_0}{2}\eta^2,$$

and therefore

$$w_1(y)^2 = \frac{\mu^2 - \eta^2}{a} y + \eta^2. \quad (25)$$

Since w_1 must be negative, we get

$$w_1(y) = -\sqrt{\frac{\mu^2 - \eta^2}{a} y + \eta^2}, \quad y \in [0, a],$$

and

$$\sigma_1(y) = -p_0 \sqrt{\frac{\mu^2 - \eta^2}{a} y + \eta^2} + \frac{(\mu^2 - \eta^2)^2}{4a^2}, \quad y \in [0, a]. \quad (26)$$

Remark 1. Since $w_1'' > 0$ then w_1 is convex.

Remark 2. Since $\sigma_1(0) < 0$ we have that $h_1 \in V_1$ (hence $h_2 \in V_2$), otherwise, as already remarked after (13), the last equation in (17) cannot be satisfied.

Finally, by (18)₁ we have

$$\begin{aligned}\bar{\sigma}_r &= \sigma_1(a)w_1'(a) - \beta_2 \\ &= \frac{p_0}{2} \left(\frac{\mu^2 - \eta^2}{2a^2\mu^2} \sqrt{4a^2(\mu^2 + \eta^2) + (\mu^2 - \eta^2)^2} + \frac{\mu^2}{\lambda - a} \right).\end{aligned}$$

The previous analysis delivers a family of curves depending on three parameters μ , λ and η . Some curves of this family for a fixed μ are depicted in Figure 5.

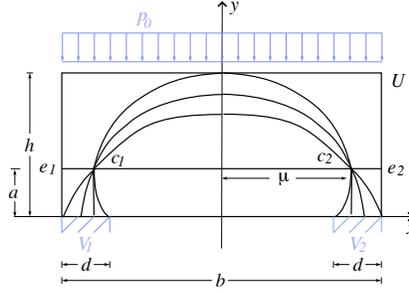


Figure 5: some singular curves for fixed μ

We notice that the curves C_1 and C_3 can be either convex or concave. In Figure 6 are depicted the line tensions in the bulk material in both cases. According to (15), in Case 1 we have a tension $-p_0$ also in the region delimited by the points a , b and c ; even if the solution we found is equilibrated, from a physical point of view in such a region it would be more natural to take $\mathbb{T}_a = 0$. This assumption, though, does not fit with our requirement that \mathbb{T}_a is smooth in $U \setminus (C_u \cup C_r)$. We therefore believe that, physically, the solution depicted in Case 2 should be preferred.

Hereafter, we call *optimal singular curve* the curve that minimizes the maximum stress in the no-tension material. According to (23) and (26), the maximum in the curve C_2 is given by

$$\max \sigma_2 = -p_0 \sqrt{\frac{\mu^4}{4(\lambda - a)^2} + \mu^2},$$

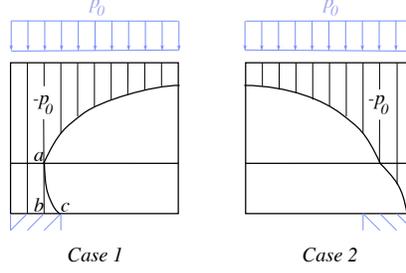


Figure 6: line tensions for different curves C_1 and C_3

while the maximum in the curve C_1 is

$$\max \sigma_1 = \begin{cases} -p_0 \sqrt{\frac{\mu^2 - \eta^2}{a} a + \eta^2 + \frac{(\mu^2 - \eta^2)^2}{4a^2}} & \text{if } \mu > \eta, \\ -p_0 \sqrt{\eta^2 + \frac{(\mu^2 - \eta^2)^2}{4a^2}} & \text{if } \mu \leq \eta. \end{cases}$$

Since to find the optimal singular curve we have to minimize the above maximum stresses, it is convenient to take $\mu \leq \eta$ and $\lambda = h$. Thus, the optimal singular curve is defined for $\lambda = h$ and for μ and η that minimize the function

$$f(\mu, \eta) := \max \{f_1(\mu, \eta), f_2(\mu)\}$$

with

$$f_1(\mu, \eta) := -p_0 \sqrt{\eta^2 + \frac{(\mu^2 - \eta^2)^2}{4a^2}},$$

$$f_2(\mu) := -p_0 \sqrt{\frac{\mu^4}{4(h-a)^2} + \mu^2},$$

on the set

$$D := \{(\mu, \eta) \in \mathbb{R}^2 : 0 \leq \mu \leq \eta, b/2 - d \leq \eta \leq b/2\}.$$

To write f explicitly we study the sign of the function $g(\mu, \eta) := f_1(\mu, \eta) - f_2(\mu)$. We find that $g(\mu, \eta) = 0$ on a monotone increasing curve $\mu = h(\eta)$ intersecting the segments $(0, b/2 - d) \times \{b/2 - d\}$ and $(0, b/2) \times \{b/2\}$ as depicted in Figure 7. On the left of the curve $\mu = h(\eta)$ the function f is equal to f_1 while on the right it is equal to f_2 . A direct computation shows that the minimum is achieved in the point denoted by A in Figure 7. The minimum is therefore achieved for $\eta = b/2 - d$ and for μ strictly less than η .

Of course, as a approaches zero, μ approaches η .

The optimal singular curve, which is represented in Figure 8, resembles the line tensions found experimentally on reinforced concrete beams, see Park and Paulay [13, Figure 7.8].

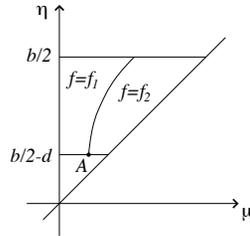


Figure 7: the domain D



Figure 8: the optimal singular curve for a slender panel

4. Conclusions

A theory for panels of no-tension material with a reinforcing tensile resistant unidimensional material has been put forward. Following the work by Lucchesi, Silhavy and Zani [8], we have allowed the stresses to be singular, in that they may concentrate on curves. The stress field in the no-tension material has been assumed to be negative semi-definite, while on the reinforcement the stress has been taken to be positive. In particular, the equilibrium equations for a rectangular panel loaded on the top by a uniform vertical load and with an horizontal reinforcement have been studied and solved. The family of solutions that we have found can be parametrized by means of singular curves where the stress in the no-tension material concentrates. Among these curves we determine the curve that minimizes the maximum stress: this singular curve resembles the line tensions found experimentally on reinforced concrete beams.

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